

PREDICTION OF AERODYNAMIC PERFORMANCE OF AN AFTERMARKET VORTEX STABILIZER

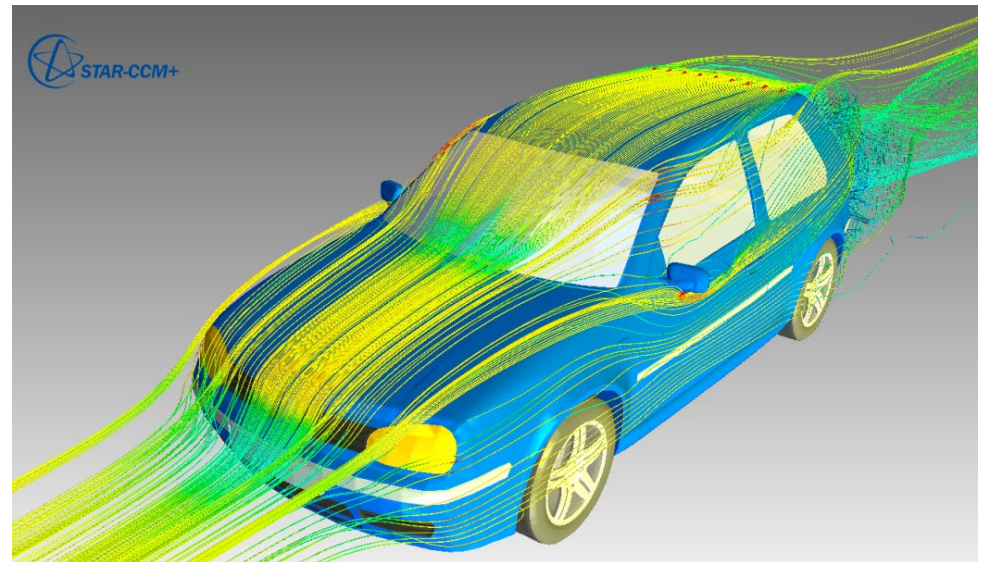
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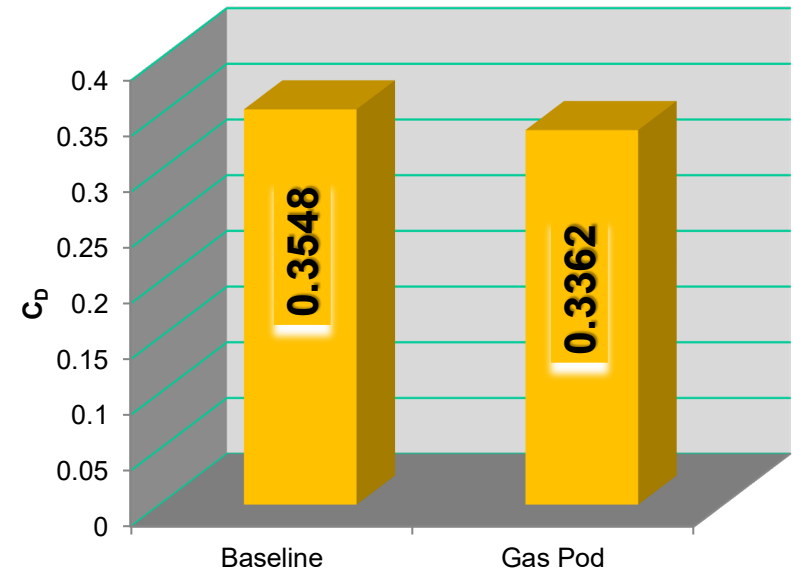
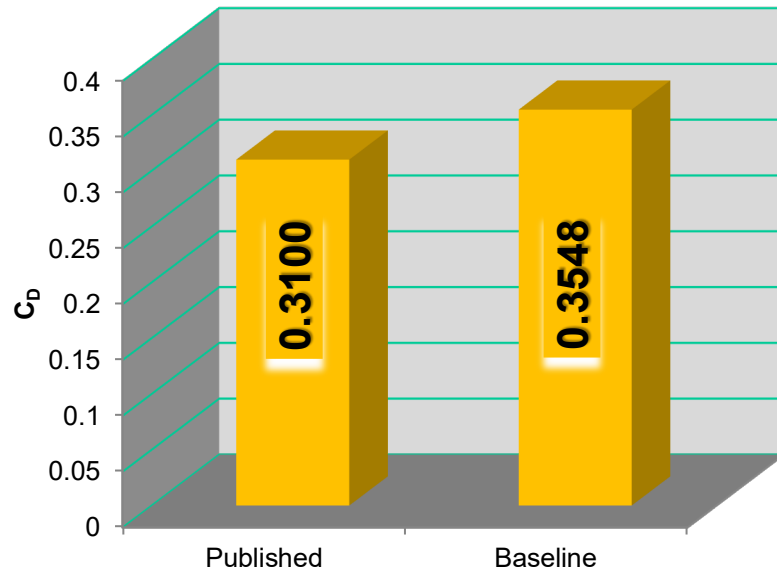
Disclaimer

CD-adapco's engineering analysis Results provided herein are dependent on the accuracy of the geometry, boundary conditions, material properties, and other data provided by your Company and may deviate from real physical operation. These Results are not intended as a substitute for sound engineering judgment and CD-adapco is not liable for the outcome of the use of these Results.

Model Overview

- The objective of this project was to investigate the effect of vortex stabilizers installed on the rear roof of a car for aerodynamic performance in terms of drag reduction.
- A 3D CAD geometry of a production personal vehicle (VW Golf IV two door) was selected and purchased from the website 3DCADBrower.com, which provides three-dimensional geometries of common vehicles. These surfaces are not provided by the OEM, and as such do not necessarily conform with the actual dimension of the vehicle.
- Two CFD models were created by **CD-adapco** using **STAR-CCM+** version 5.04.008. These models represent a baseline geometry of the vehicle, as well as one geometry with the GasPods.
- Each GasPod was positioned 4 inches apart from each other in the lateral direction and 10 inches upstream from the trailing edge of the roof.
- A half symmetry model was used to simulate the flow field.
- The computational models were generated based on standard mesh settings for external aerodynamic analyses with appropriate refinement particular to this model.
- Analyses were performed for the vehicle speed of 65 miles/hour.

Key Results: Drag Coefficient

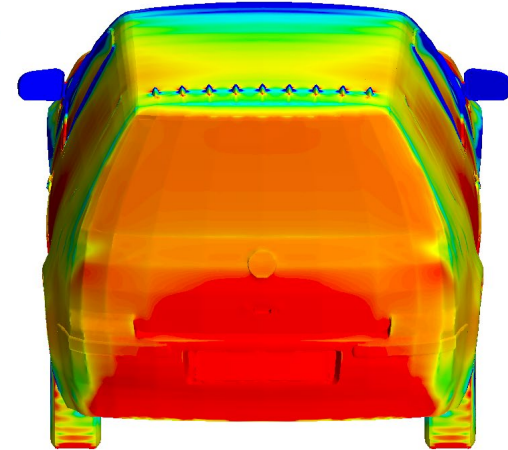
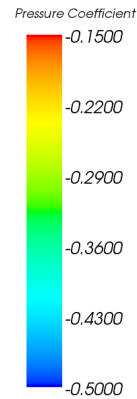
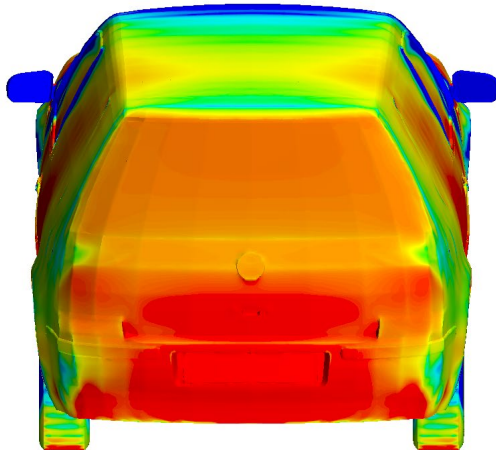
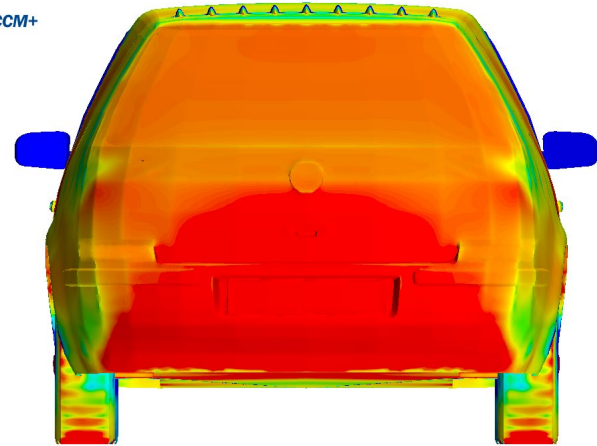
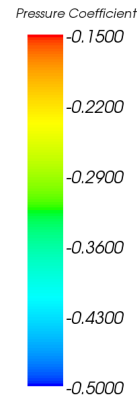
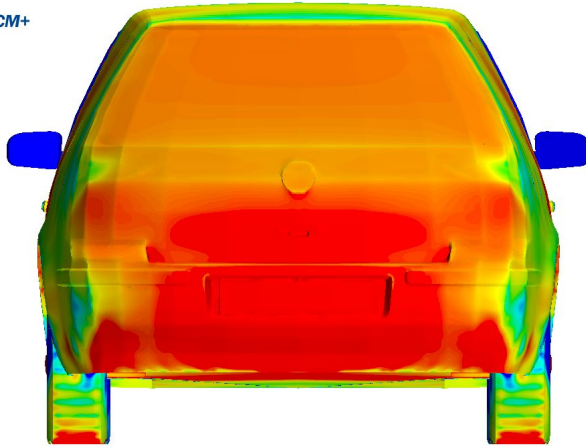


- The drag coefficient predicted from the baseline model was compared against the published data for the vehicle. (See for example <http://www.new-cars.com/2003/volkswagen/volkswagen-golf-specs.html>)
- The drag coefficient was predicted within about 13%. Considering that the CFD model is not based on actual CAD data, as well being considerably simplified (wheel wells, underbody, missing suspension and brake rotors), this is an acceptable result.
- The GasPods reduce the drag of the baseline car by about 5%.

Key Results: Base Pressure Distribution

Baseline

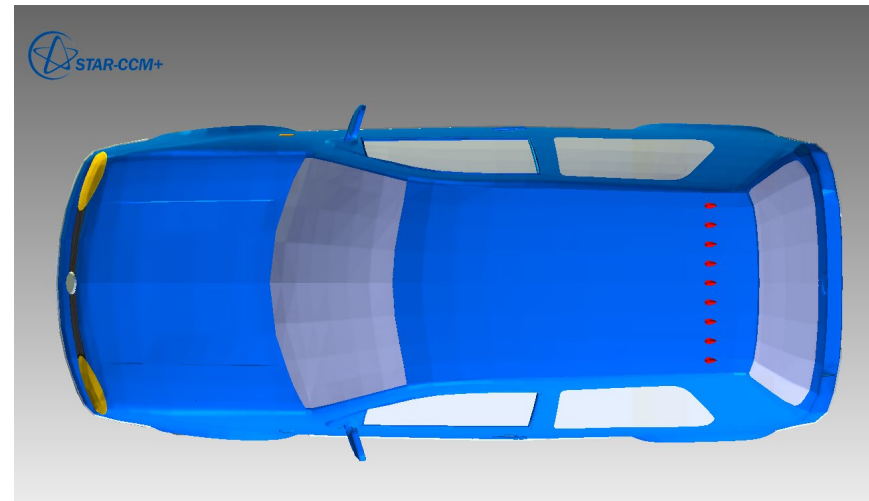
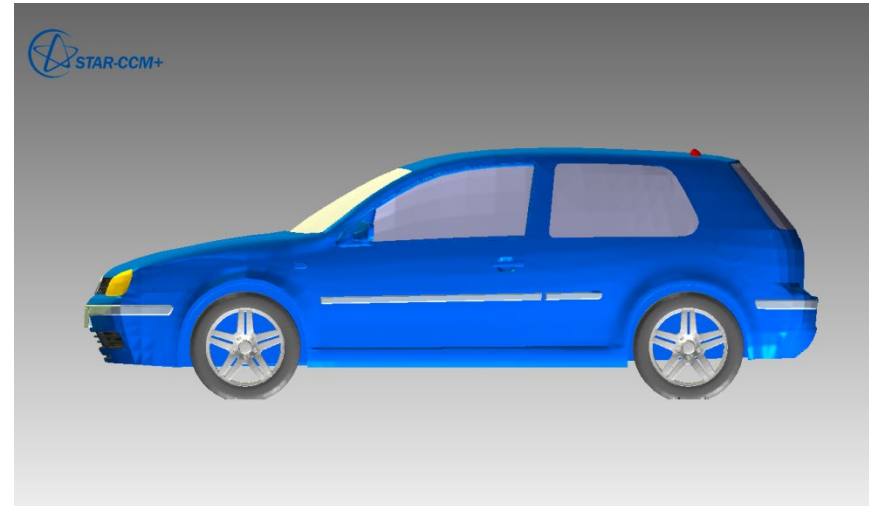
Gas Pods Installed



Summary and Discussions

- CFD results show that installing GasPods reduces the drag coefficient by about 0.0186, which is equivalent to 5.24 % reduction in drag.
- Although Gas Pods increase the profile drag, overall reduction in drag by modifying flow structures in the wake layer turns out to be greater than the addition of the profile drag.
- Presence of GasPods in the growing boundary layer decelerates the downstream flow, which results in a pressure rise inside of the shear layer.
- This pressure rise propagates through the shear layer into the base region of the vehicle.
- Presence of GasPods on the roof causes increase in base pressure of the vehicle in the wake region, which results in reduction in drag.
- This analysis demonstrates that change in the flow field, especially pressure in the vicinity of the shear layer that lies between the free stream and the re-circulation region can decrease drag on the vehicle.

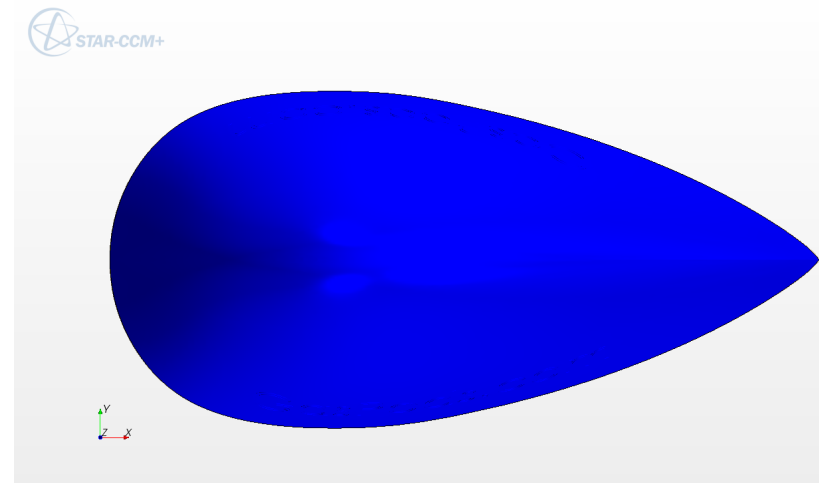
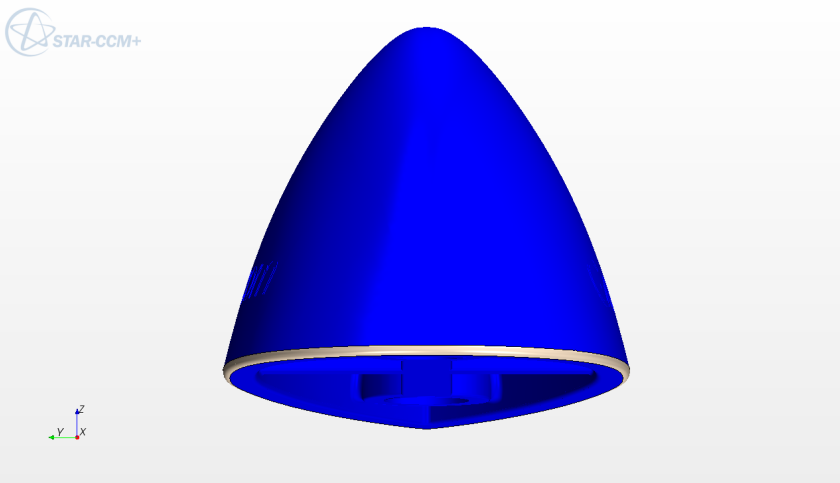
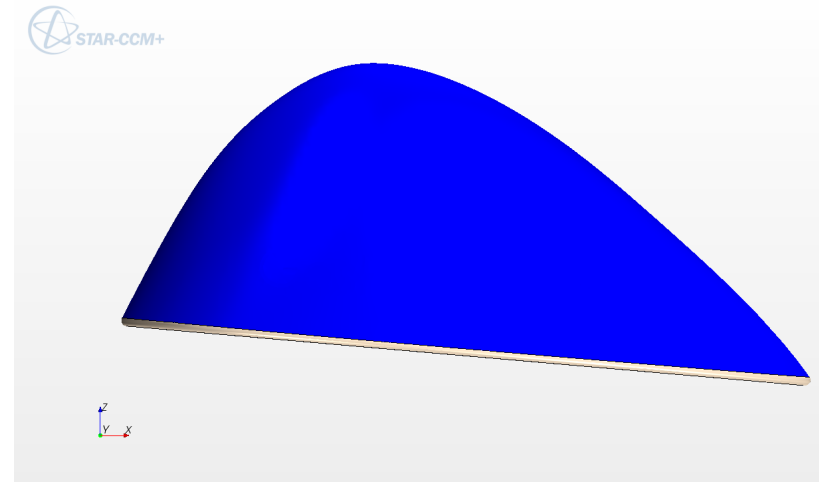
Model Overview: Selected Vehicle



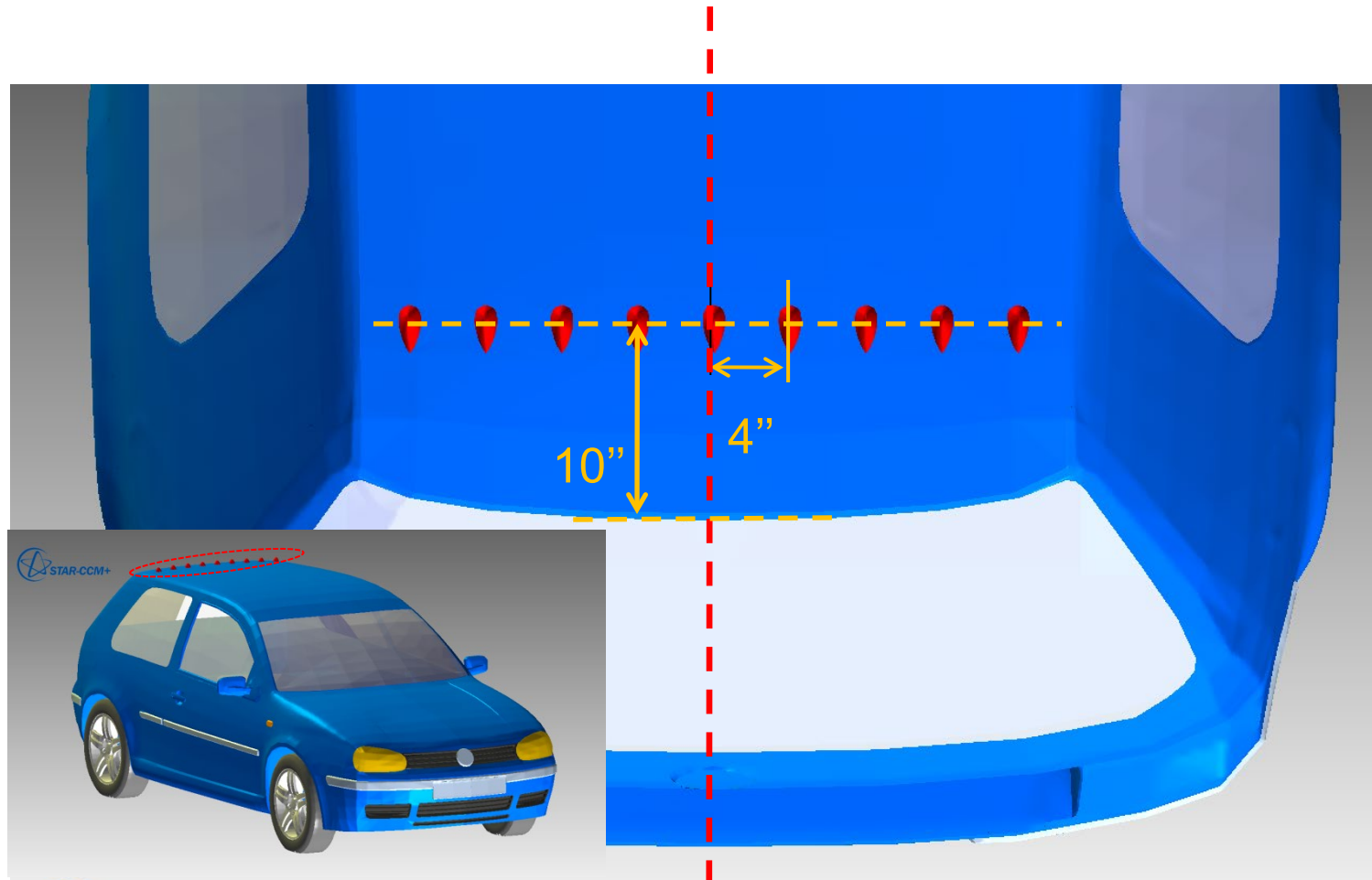
Description of Model

Model Overview: GasPod Geometry

Description of Model

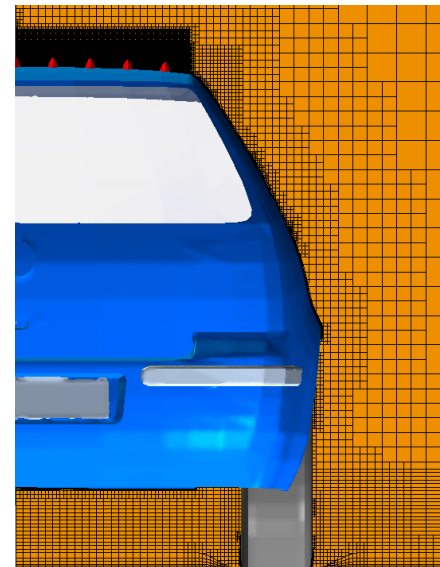
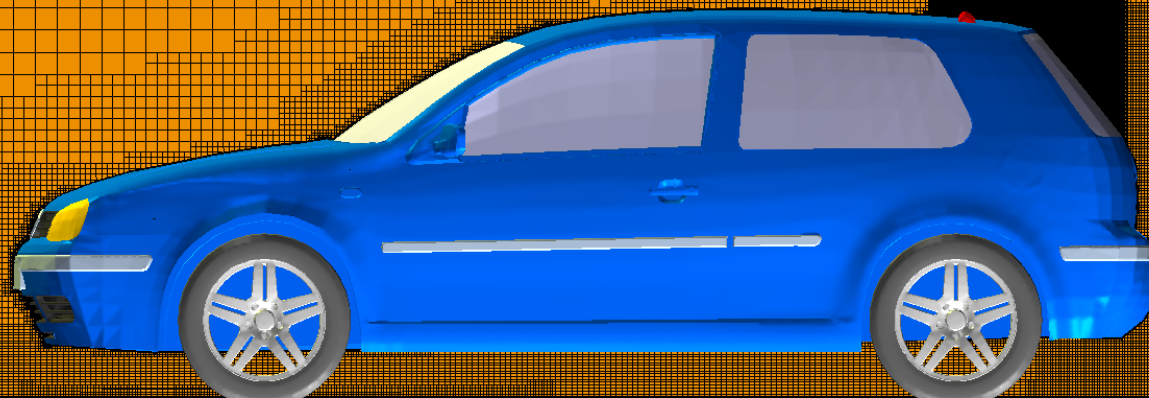
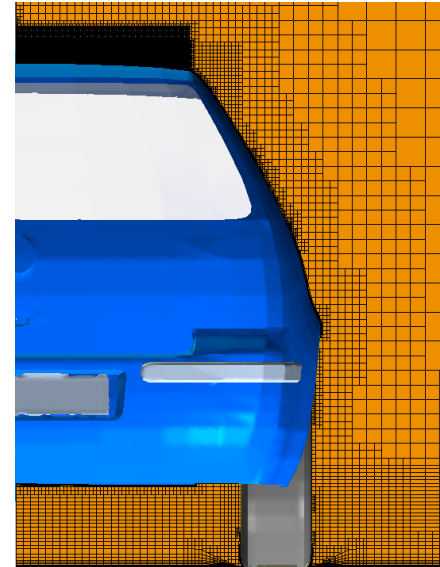
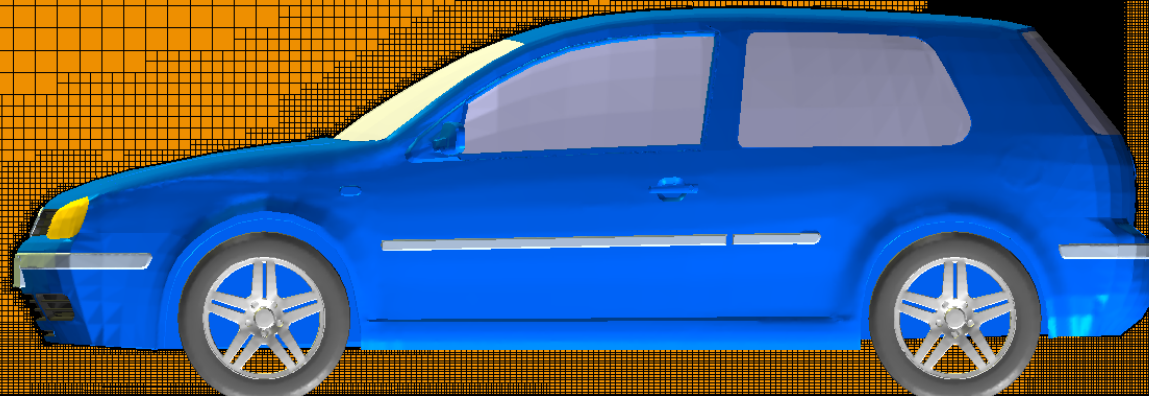


Model Overview: Position of Gas Pods



Plane of symmetry

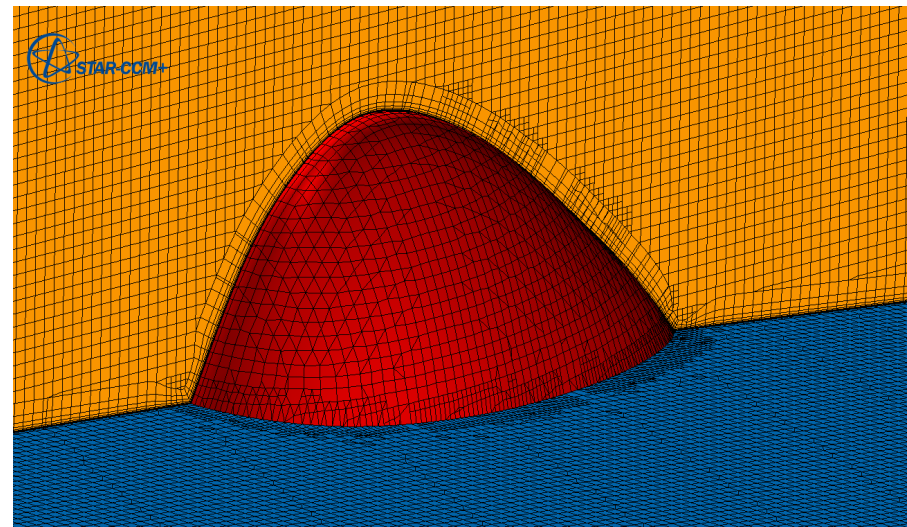
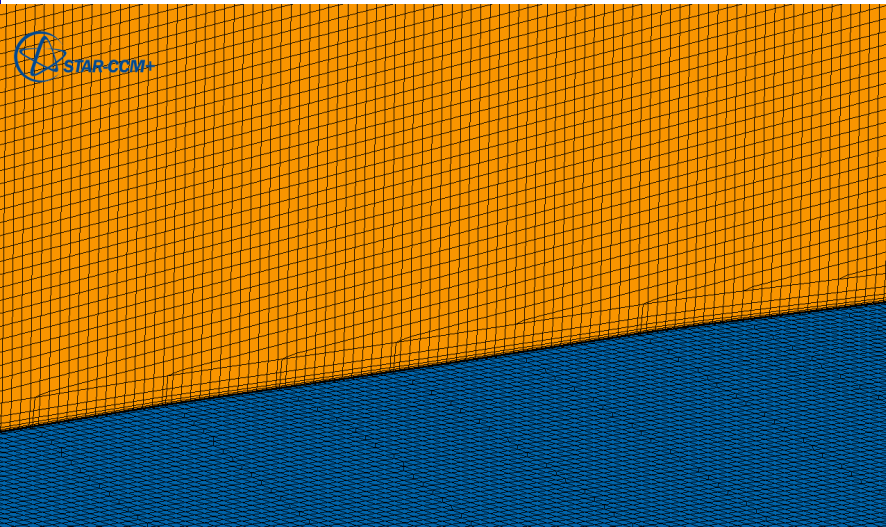
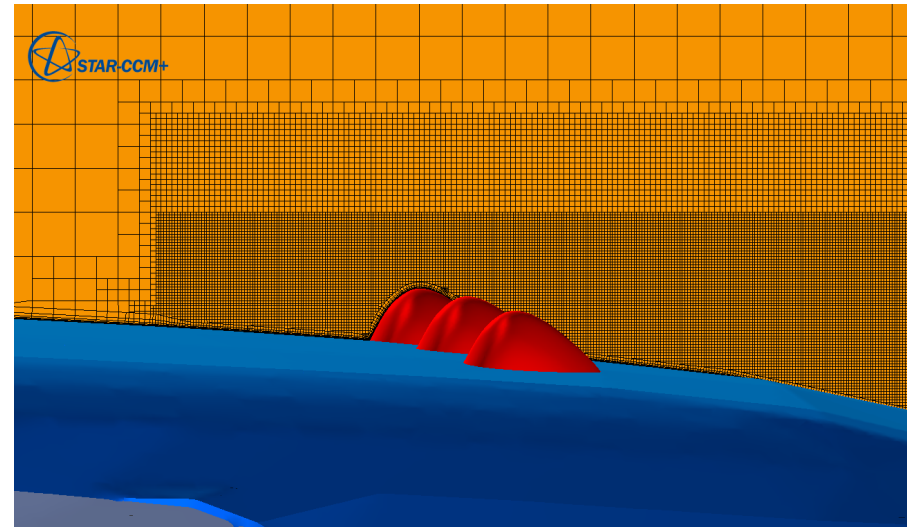
Computational Mesh



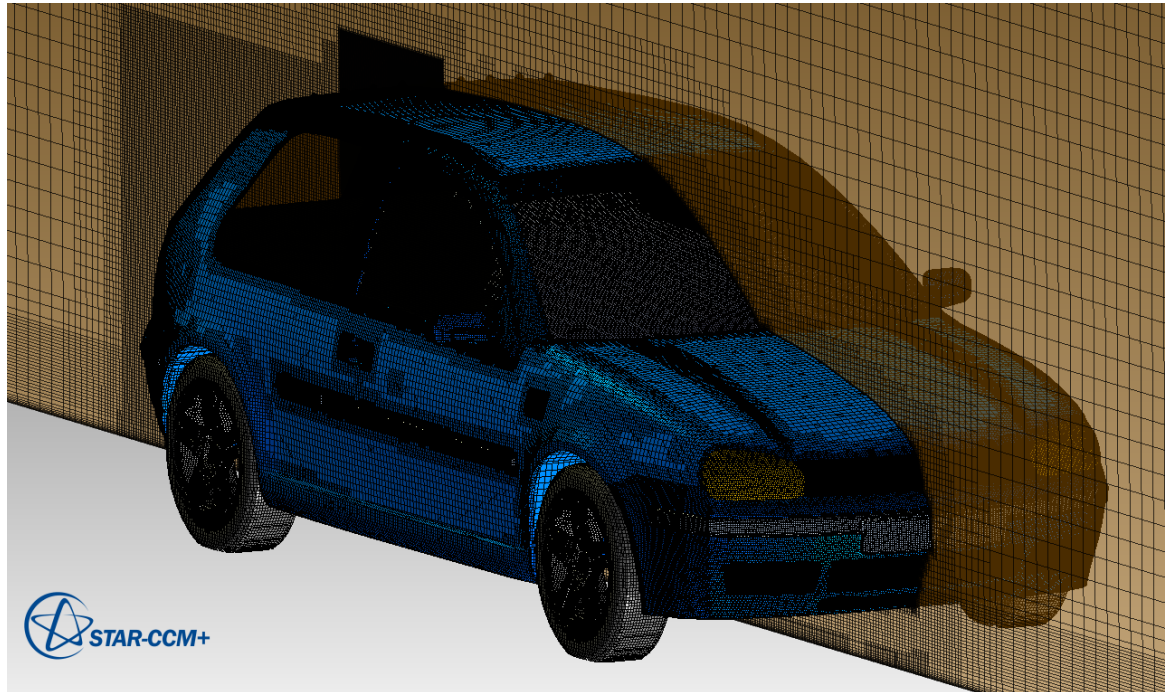
Description of Model

Computational Mesh

Description of Model



Computational Mesh

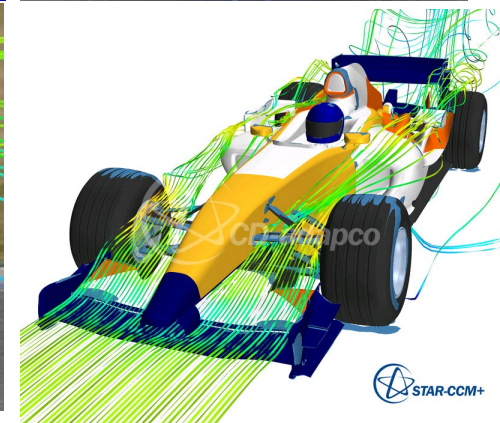
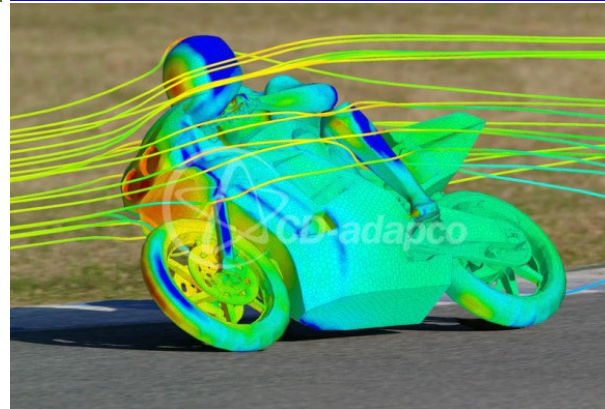
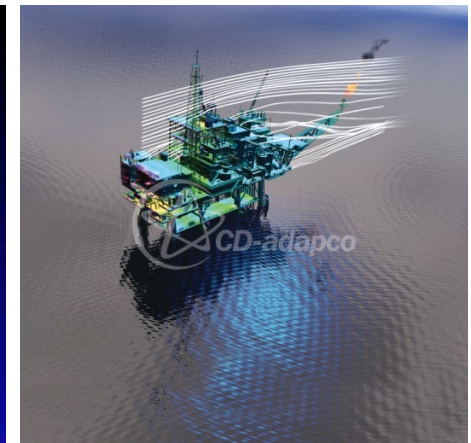
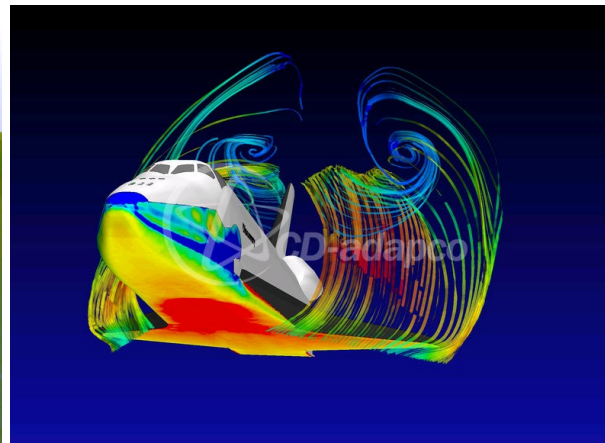
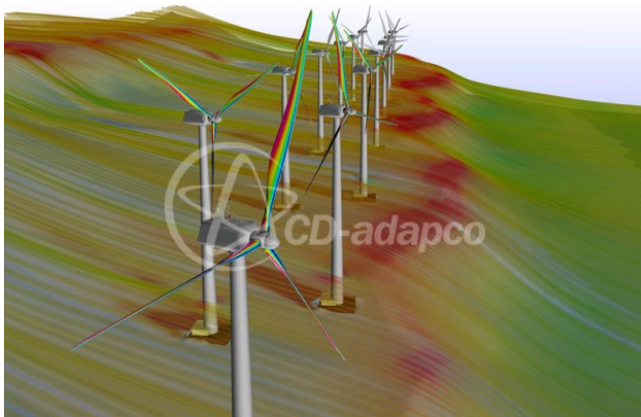


		Baseline	Gas Pods
Cell counts		23.67 millions	23.82 millions
Base size		20 mm	
Prism Layer Thickness		5.47 mm	
Near wall thickness		0.002 mm	
# of prism layers		12	
Surface size	Minimum	10 mm	
	Target	20 mm	

Computational Fluid Dynamics (CFD)



- Computational Fluid Dynamics is an analysis tool to predict physical fluid flow and heat transfer using computational methods
- CFD is widely used in engineering community to improve engineering design and to reduce the overall cost and time of product development



Drag and Drag Coefficient

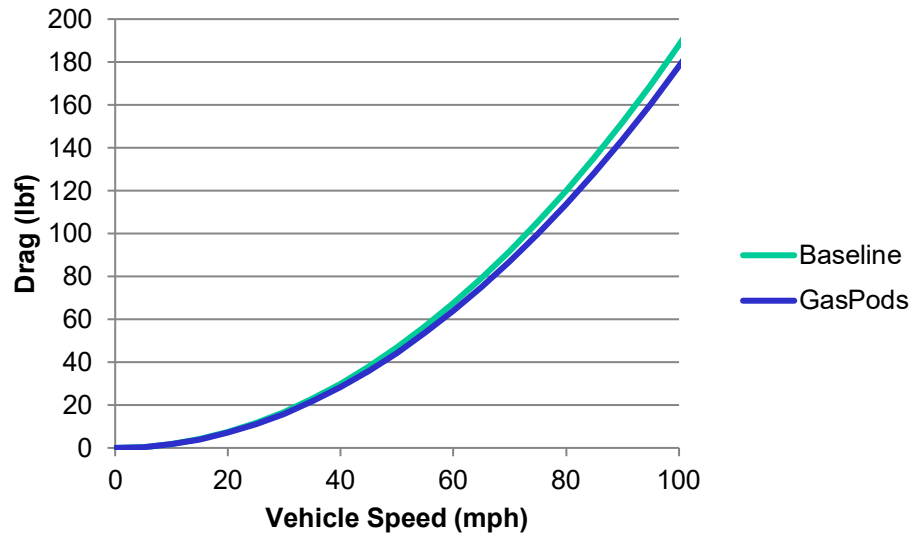
- A drag force is a component of the resultant force on an object that is parallel to the freestream.
- A drag force on a vehicle consists of two contributions: pressure (or form) drag and skin friction drag.
- Pressure drag is the stream-wise component of the pressure force integrated over the entire vehicle while skin friction drag is the stream-wise component of the shear force (friction between the air and vehicle surface) over the vehicle.
- Aerodynamic performance of various vehicles is often described by a drag coefficient which is independent of the vehicle speed and shape (area). The drag coefficient of a vehicle remains close to constant over the range of typical highway speeds (45 mph to more than 100 mph)
- A drag coefficient is defined by the following formulae

$$C_D = \frac{D}{\frac{1}{2} \rho_{ref} V_{ref}^2 A_{ref}}$$

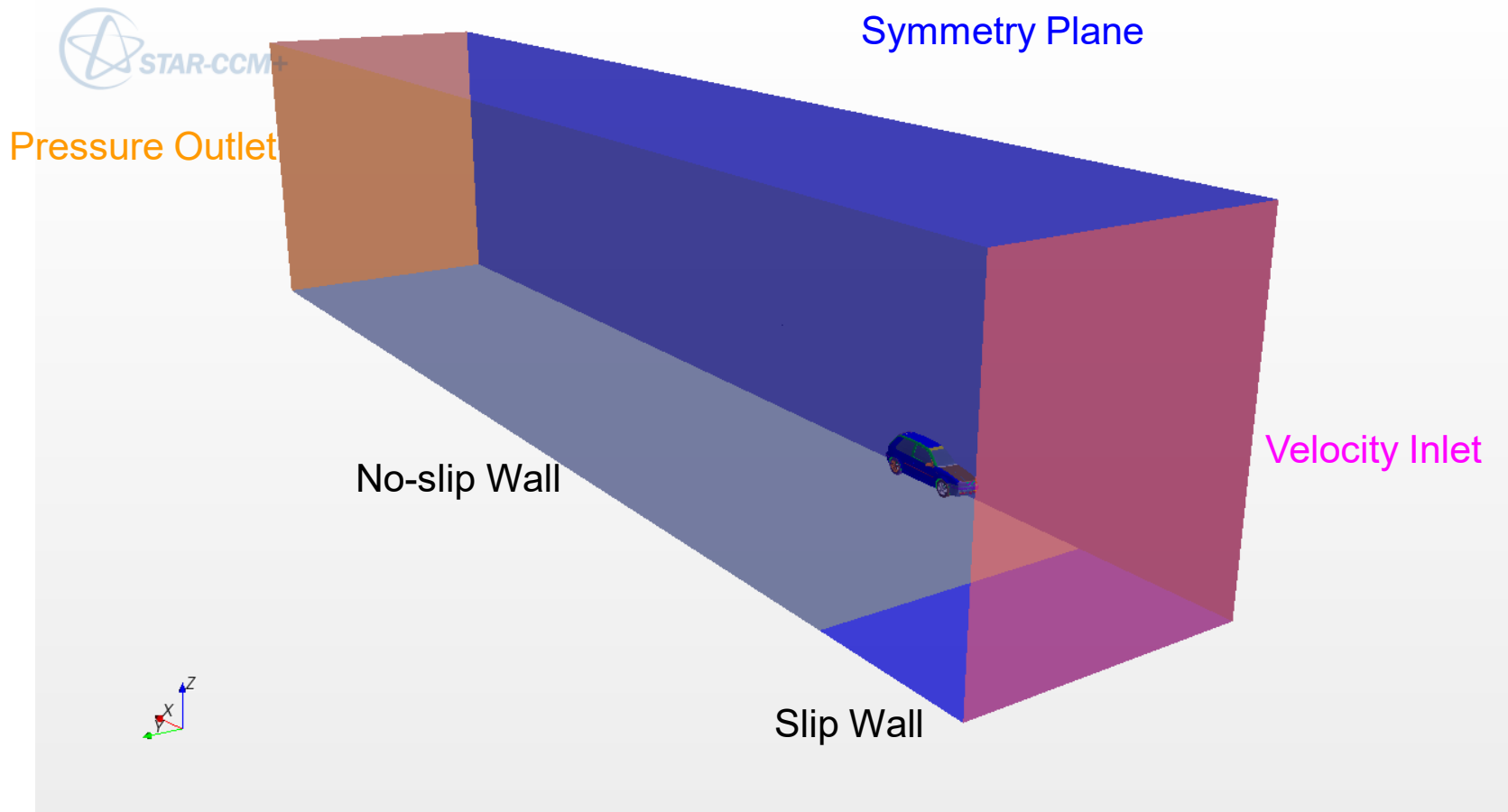
C_D	Drag coefficient
D	Drag force
ρ_{ref}	Reference density
V_{ref}	Reference velocity
A_{ref}	Reference Area (Frontal area of the vehicle)

Drag and Drag Coefficient

- As an advantage of expressing a drag force in terms of a drag coefficient, comparison of aerodynamic performance can be made regardless of the vehicle size, vehicle speed, and altitude at which the vehicle is going to be operated
- Since a drag coefficient remains relatively constant in a typical range of operations, a drag force varies with density (altitude), frontal area of the vehicle, and vehicle speed as shown below
- For a certain vehicle model at a certain altitude, a drag varies only with vehicle speed as shown below



Exterior Boundary Conditions



Exterior Boundary Conditions

Front Surface	Velocity Inlet	V = 65 mph P = 0.0 atm
Rear Surface	Pressure Outlet	Extrapolated P = 0.0 atm T = 308 K
Outer Surface	Symmetry	Extrapolated P = 0.0 atm T = 308 K
Ground	Wall	No-Slip
Front Tires (2x)	Moving Wall	No-Slip T = adiabatic $\omega_{wall} = 2126.28 \text{ rpm}$
Rear Tires (2x)	Moving Wall	No-Slip T = adiabatic $\omega_{wall} = 2126.28 \text{ rpm}$

Fluid Properties at Inlet

	Inlet Velocity	65 mph
AIR	Temperature	Iso-thermal
	Ambient Pressure	1.0 atm
	Dynamic Viscosity	1.86 E ⁻⁵ Pa-s
	Density	1.18 kg/m ³ (4.28 lb/in ³)

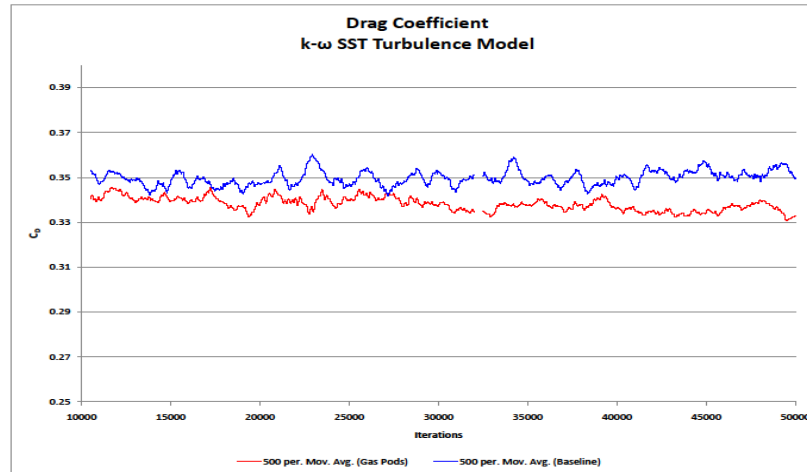
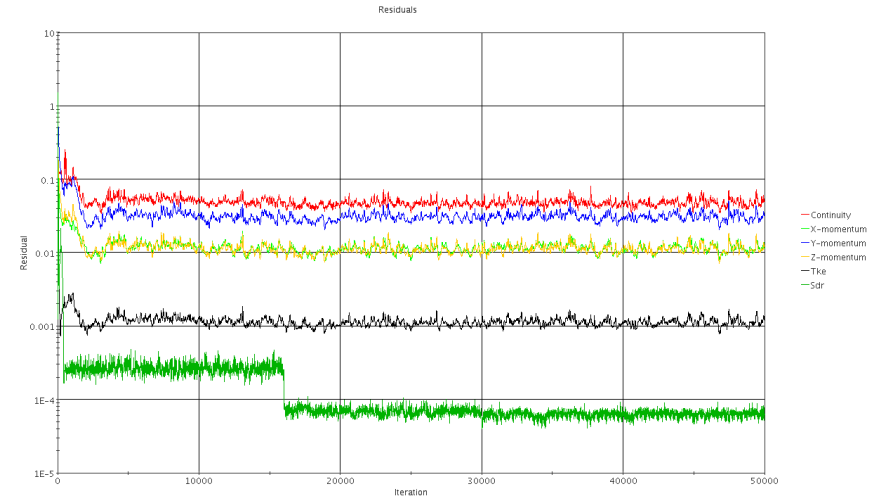
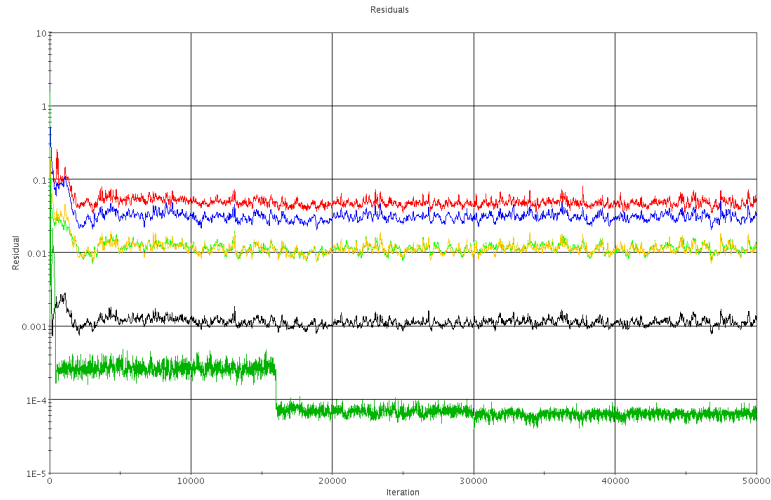
Analysis Setup, Run Statistics

Solution Algorithm		SIMPLE
Convective Flux Differencing Scheme	Velocity	Upwind (2 nd order)
	Other Variables	Upwind (2 nd order)
Turbulence Model		k- ω SST
Wall Treatment		Two layer all Y+ wall function

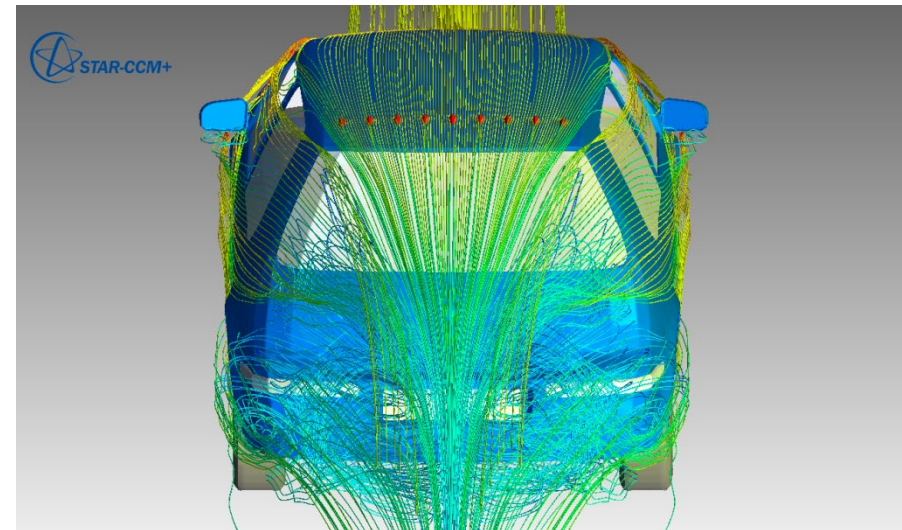
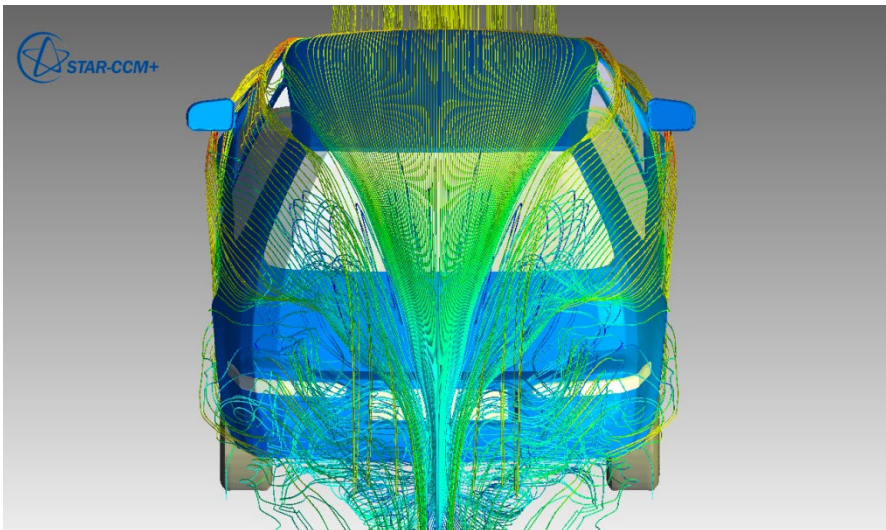
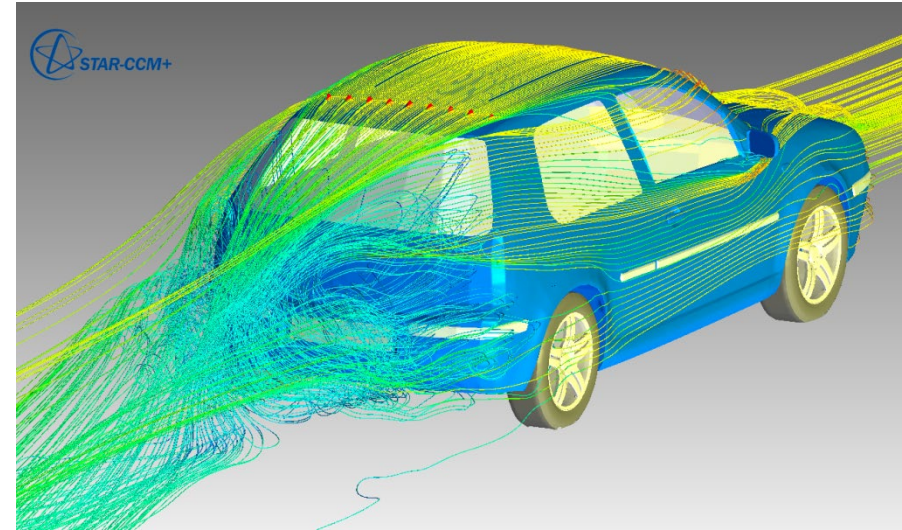
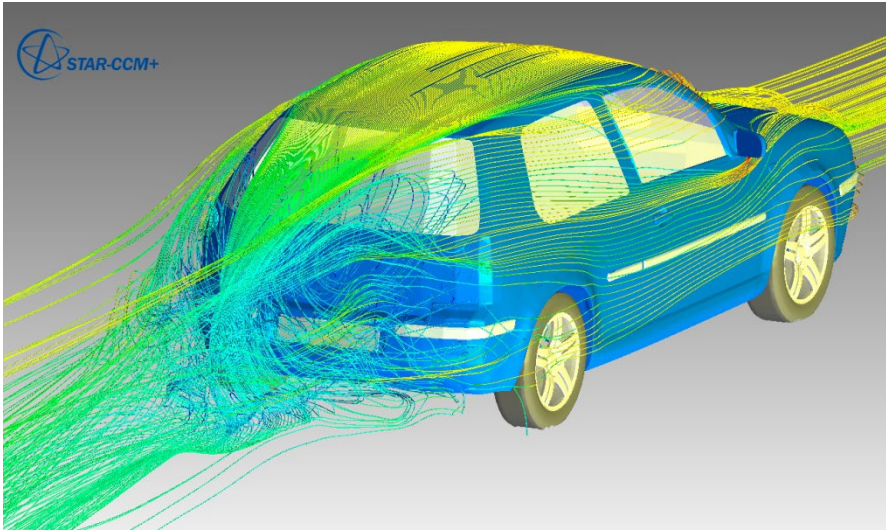
STAR CCM+ Version	5.04.008
Computing Platform	LINUX Cluster 64-bit 3.2GHz Xeon Processors
Number of Processors	32
Total number of iterations	50,000
Average Speed	0.14 sec/iteration
Total Analysis Time	8 Days 4.5 hrs
Power Usage	1.00 kW



Convergence History



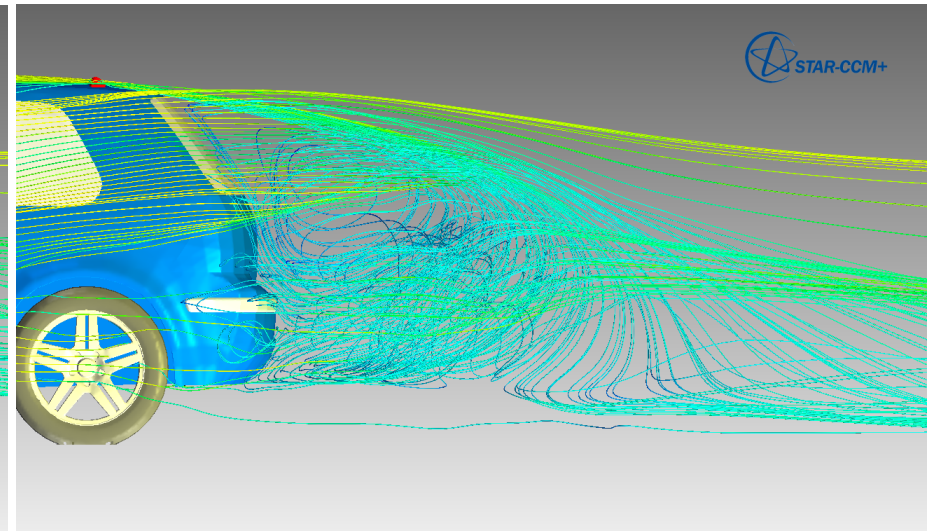
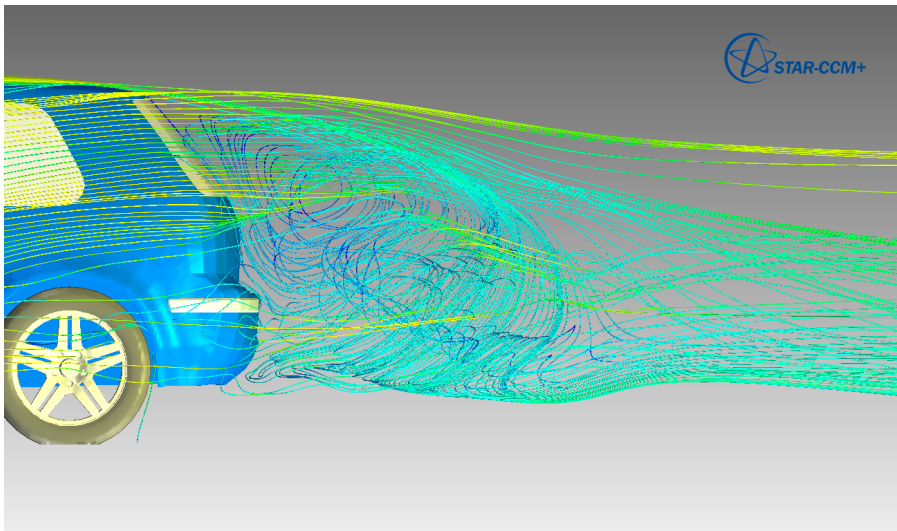
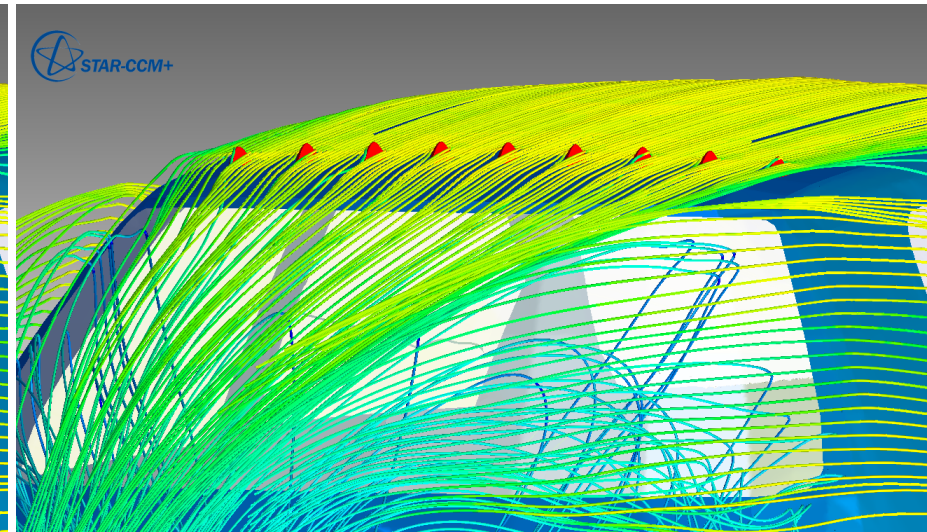
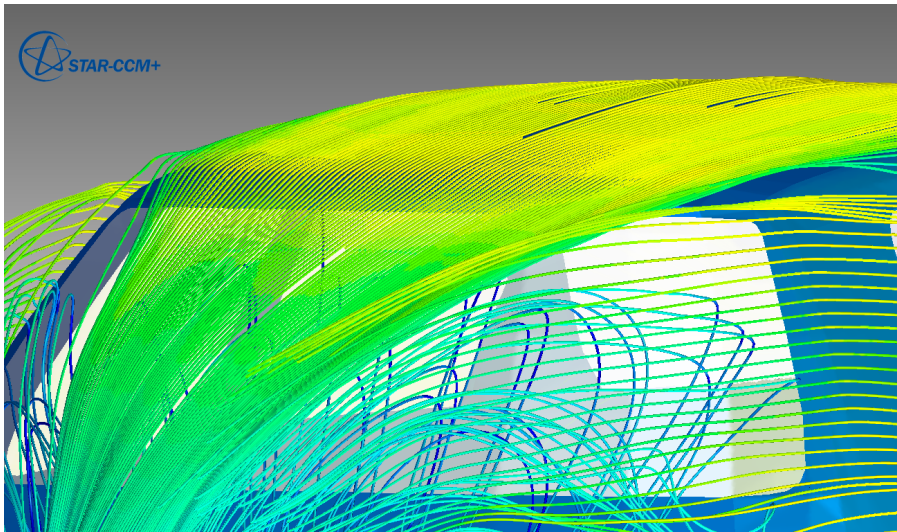
Streamlines



Results

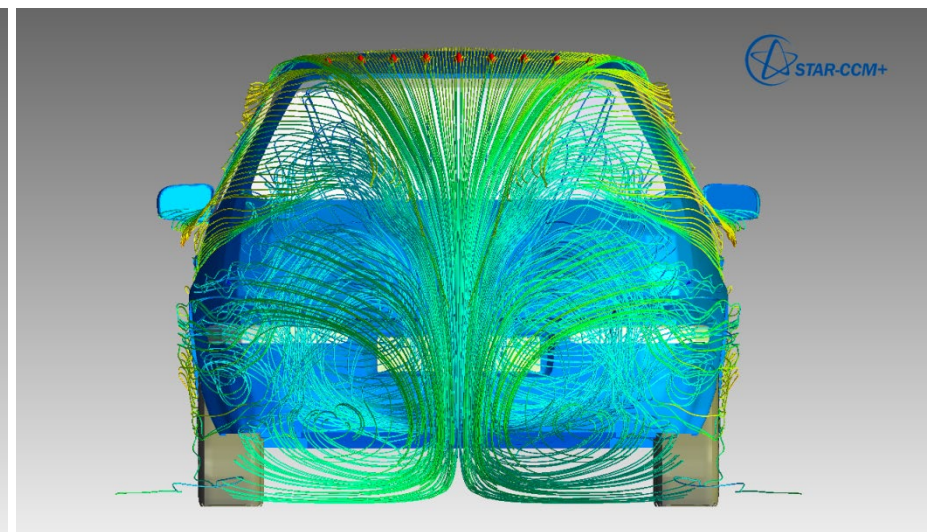
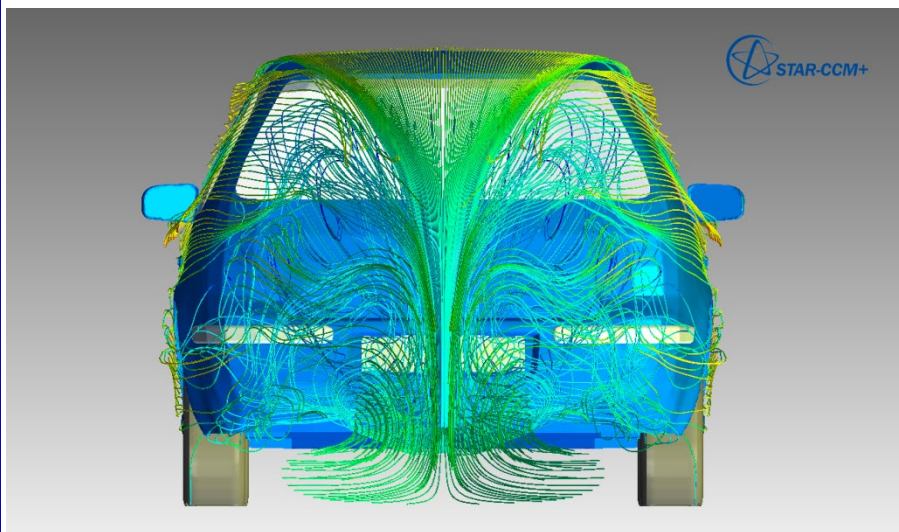
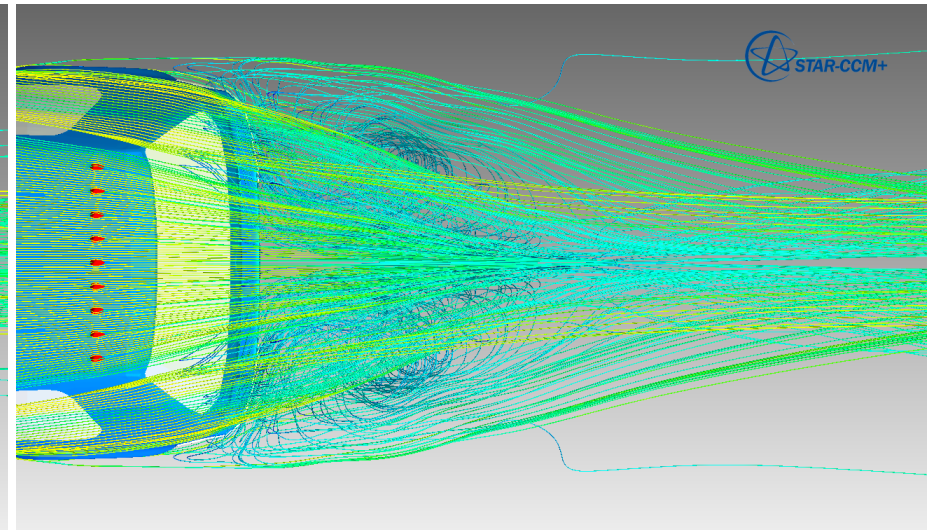
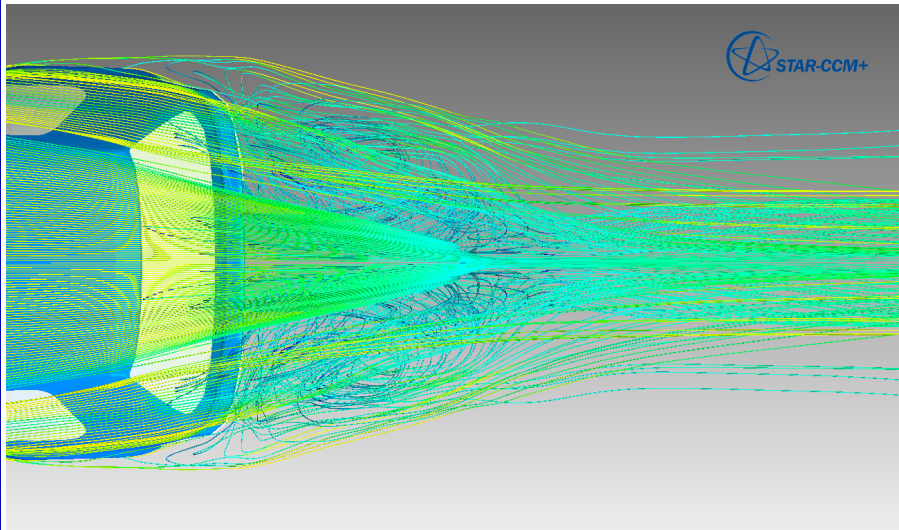
Streamlines

Results

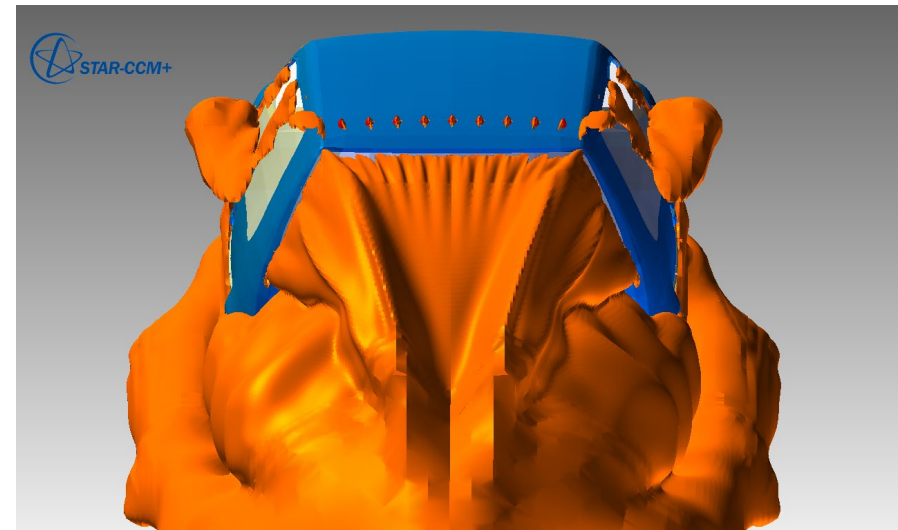
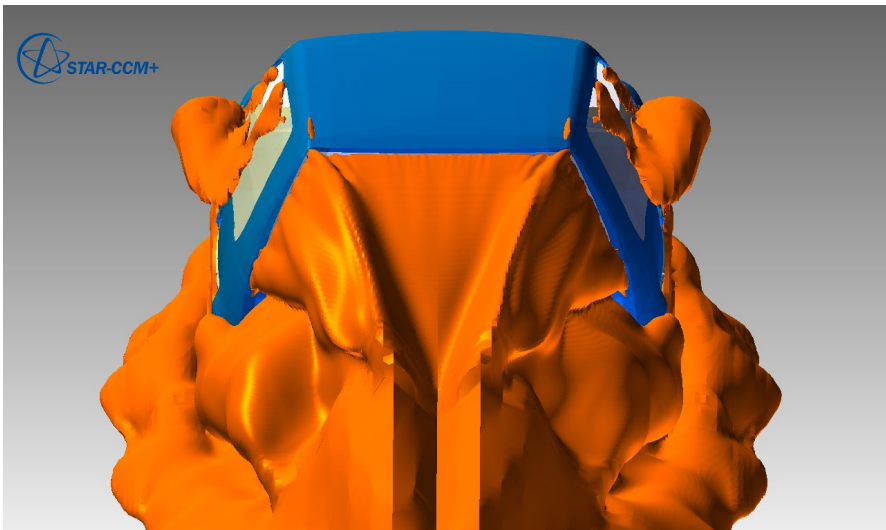
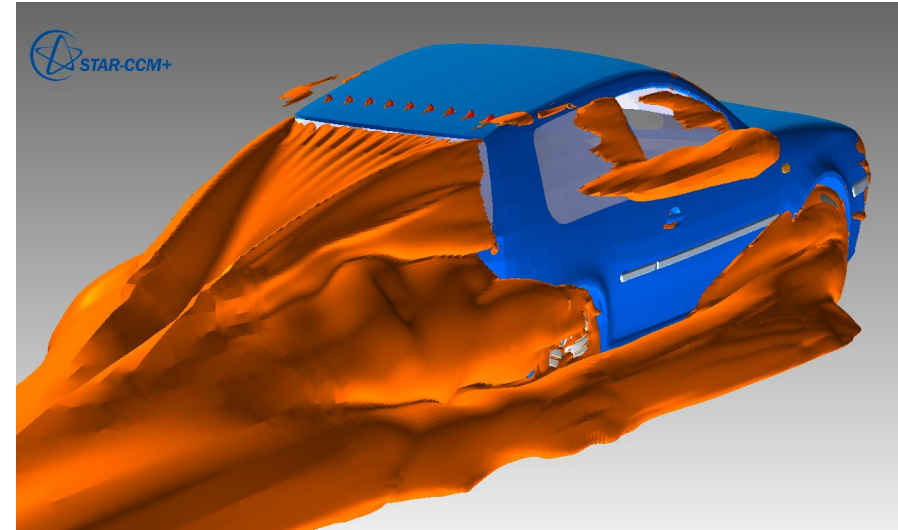
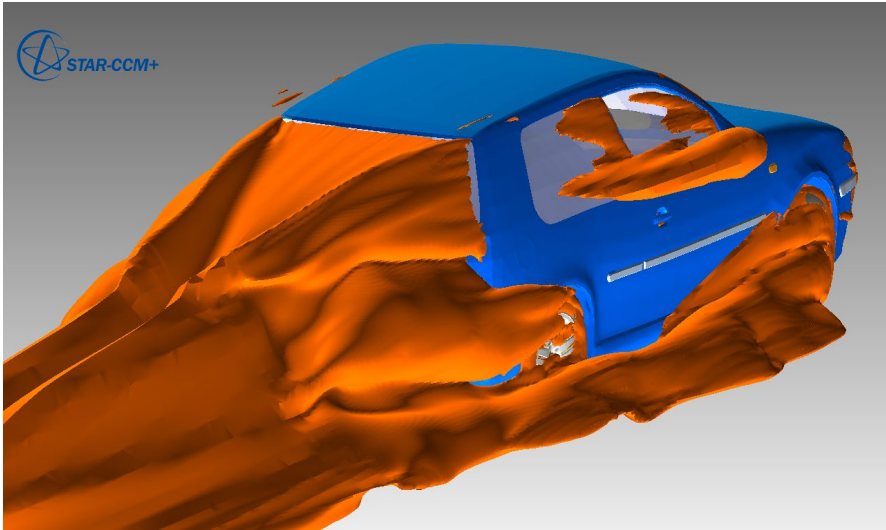


Streamlines

Results

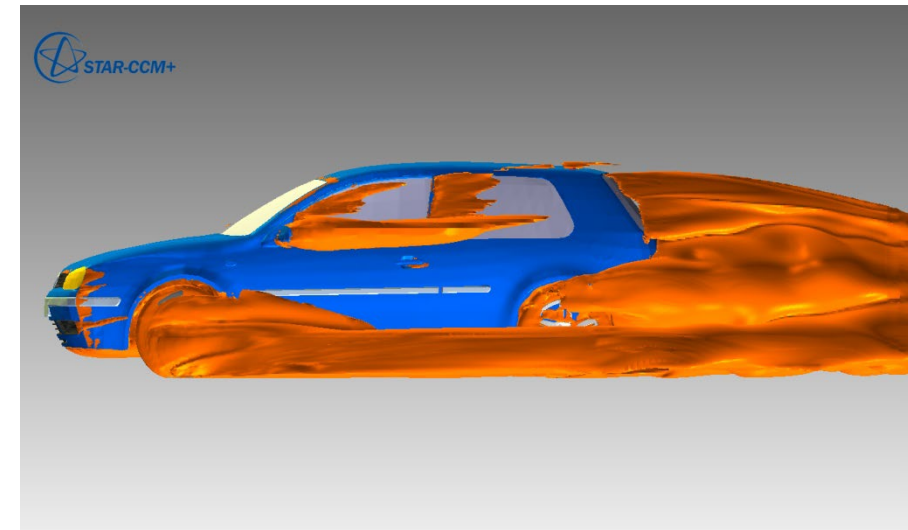
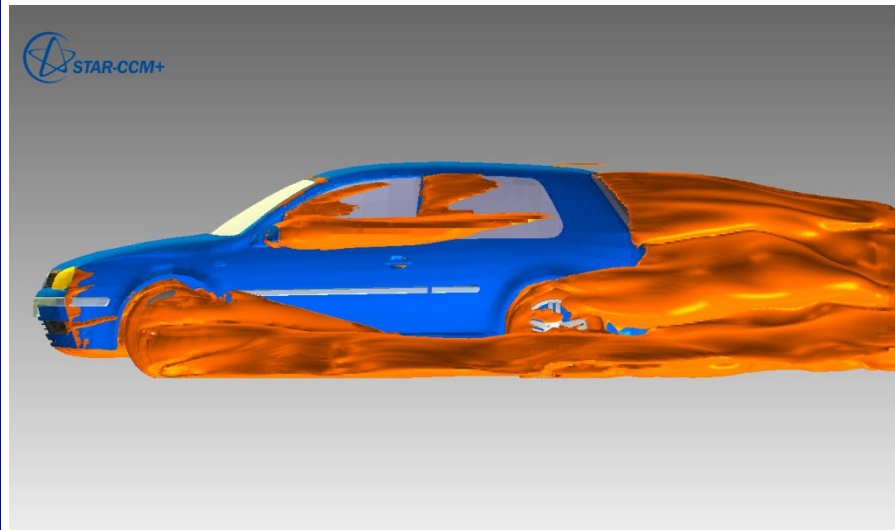
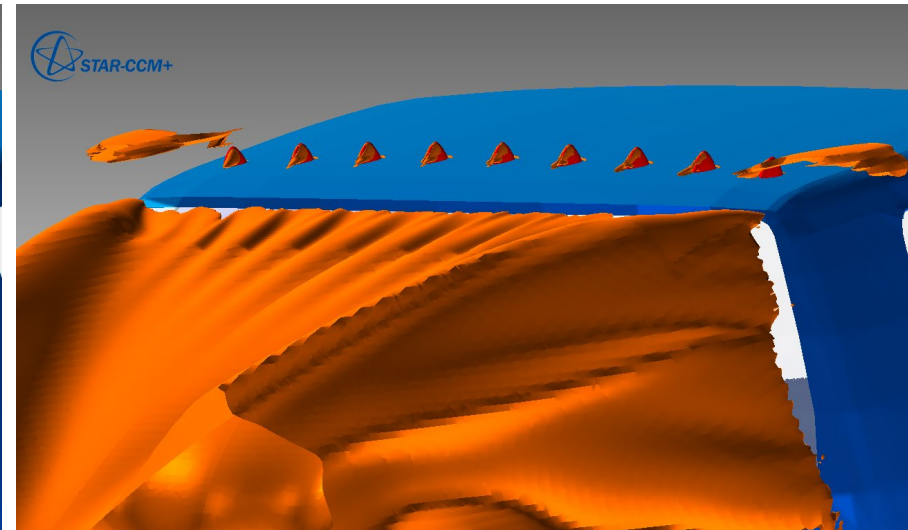
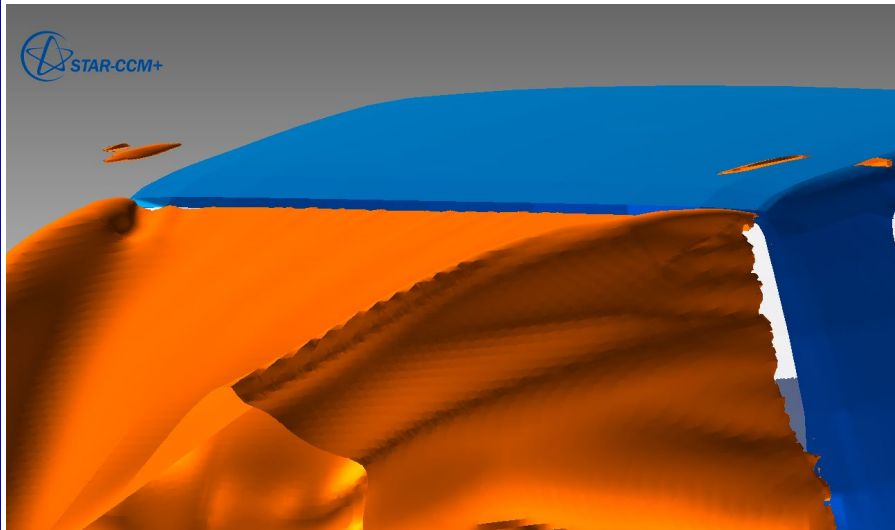


Iso-surface of Turbulence Kinetic Energy



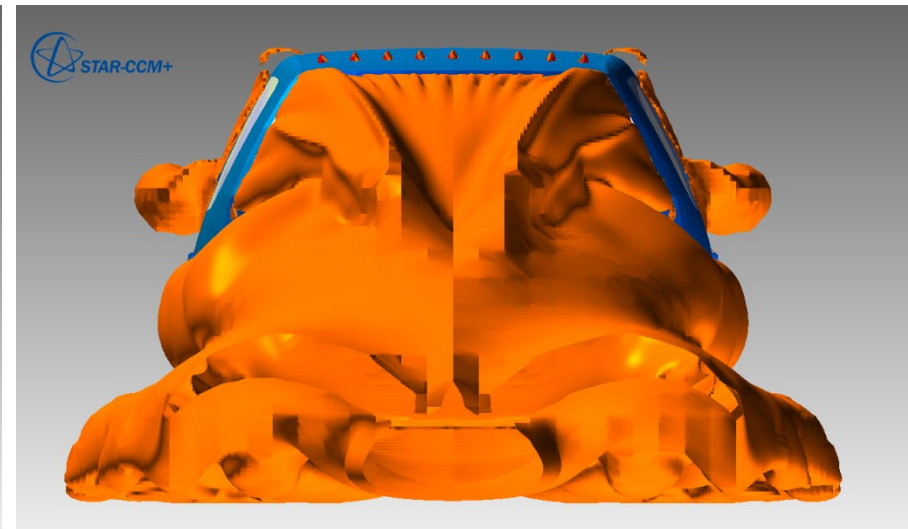
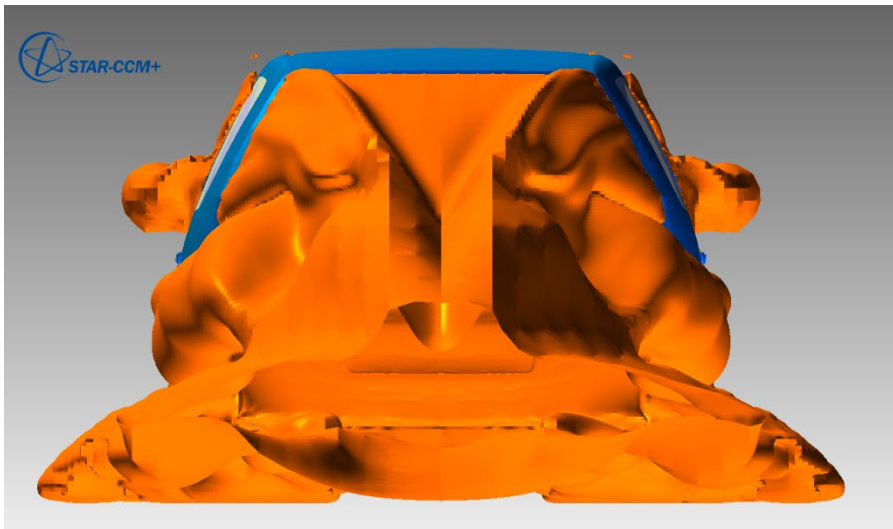
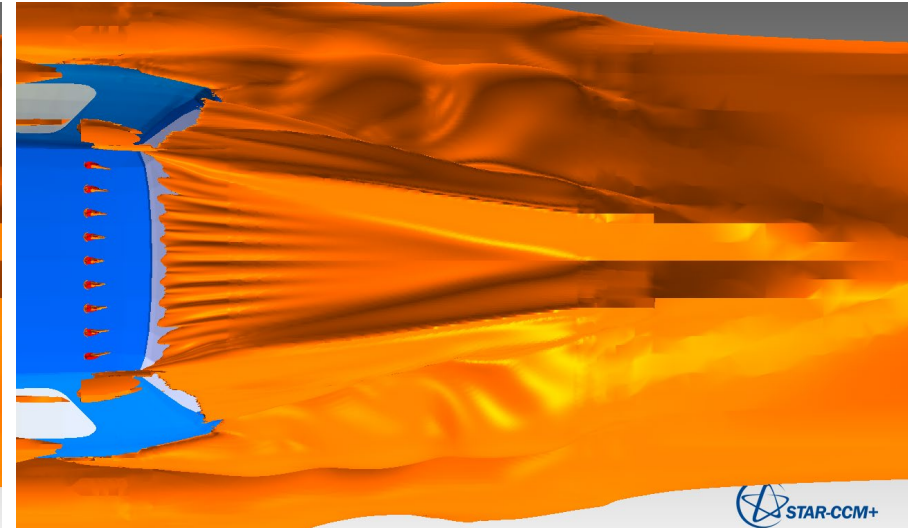
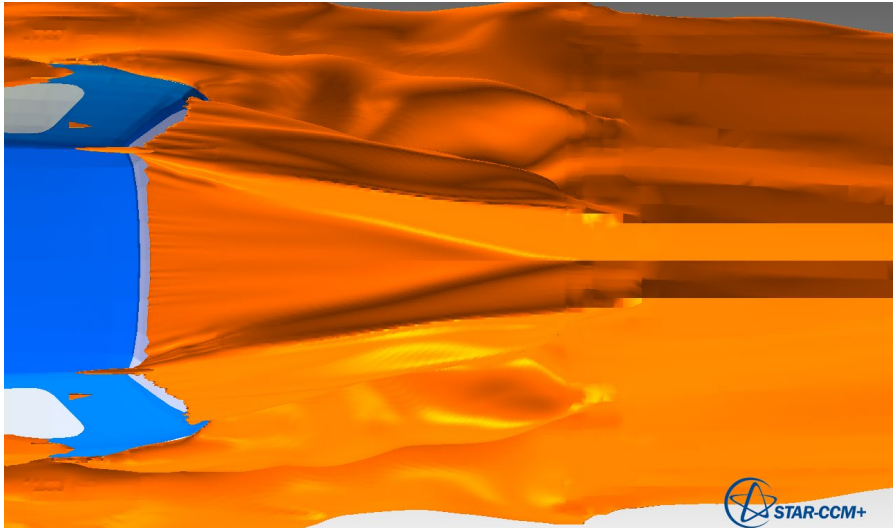
Results

Iso-surface of Turbulence Kinetic Energy



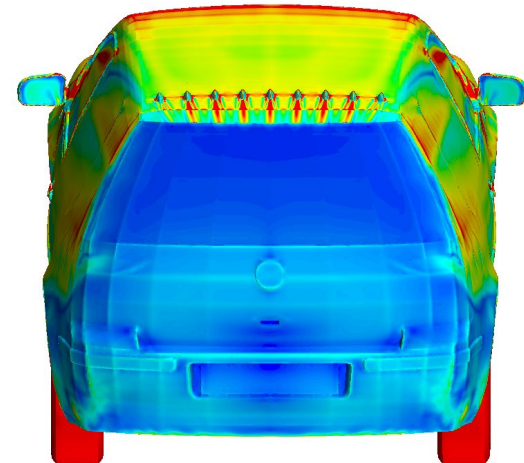
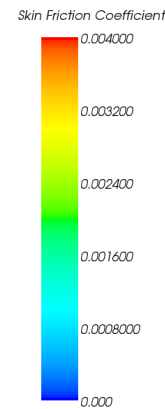
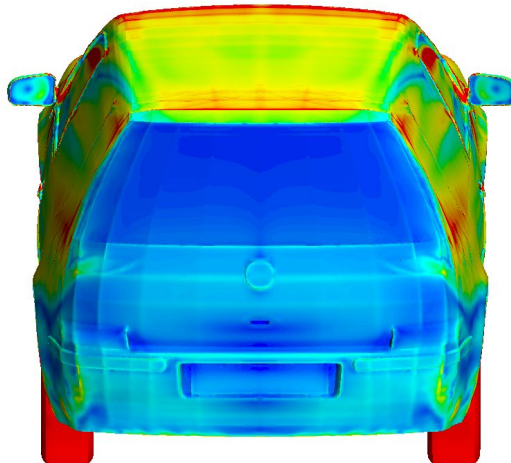
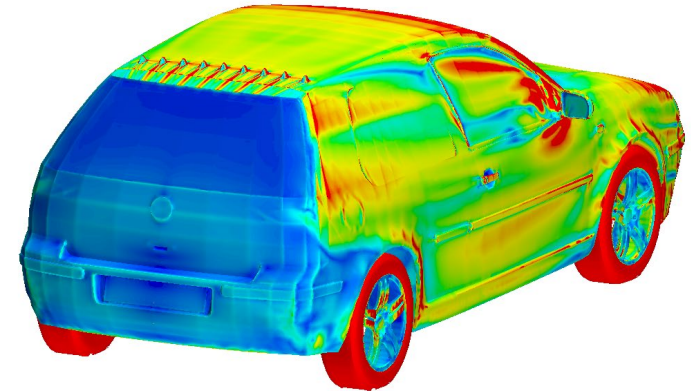
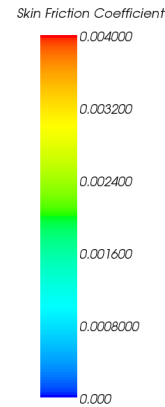
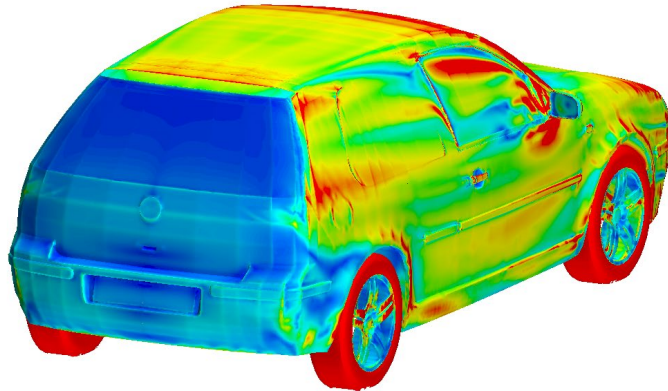
Results

Iso-surface of Turbulence Kinetic Energy



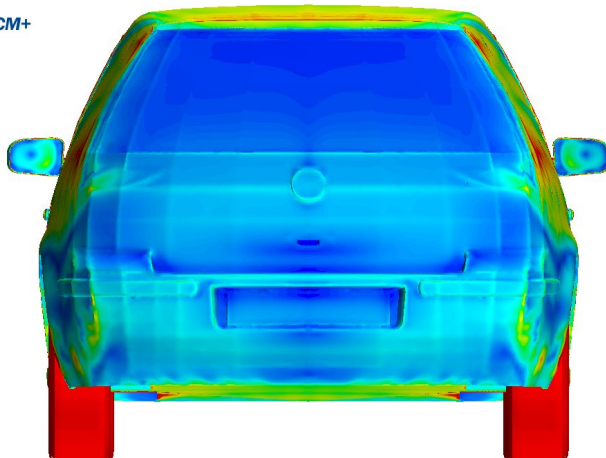
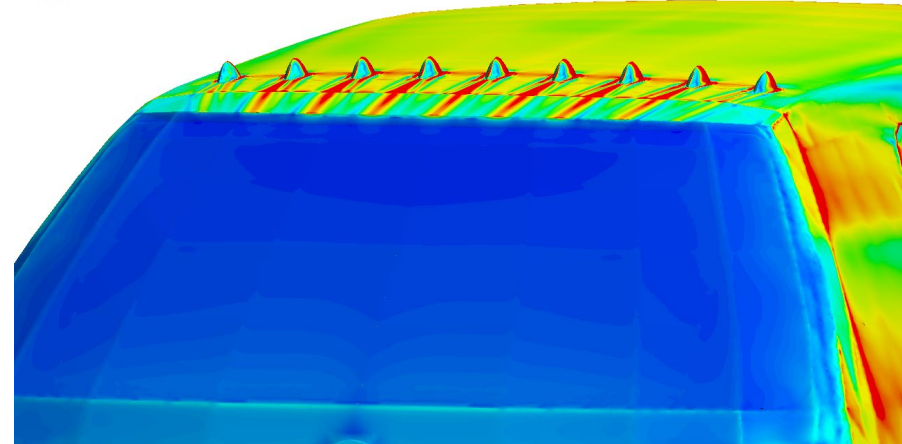
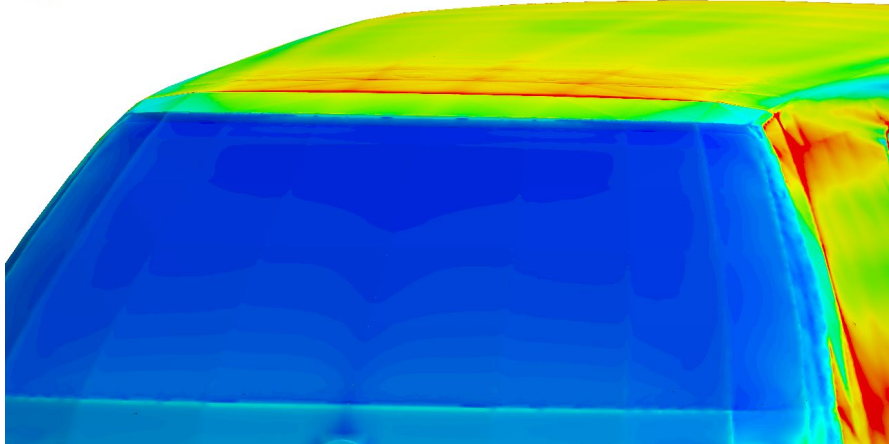
Results

Skin Friction Coefficient

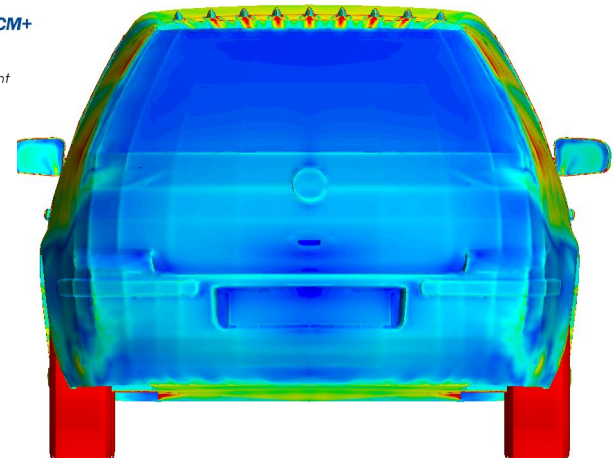
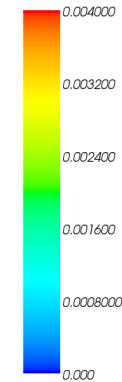


Results

Skin Friction Coefficient



Skin Friction Coefficient



Results

Definitions of Variables and Reference Values

- Pressure coefficient is another non-dimensional variable used to represent pressure (force per unit area).

$$C_P = \frac{P - P_{ref}}{\frac{1}{2} \rho_{ref} V_{ref}^2}$$

- Frontal area is the projected area of the vehicle onto the plane normal to the flow direction.
- The following values were used to convert dimensional variables such as drag, pressure, and shear stress to corresponding non-dimensional variables while reference values are arbitrary.

P_{ref}	Reference Pressure	1.0 atm
ρ_{ref}	Reference density	4.278 lb _m /in ³
V_{ref}	Reference velocity	65.0 mph
A_{ref}	Reference Area (Frontal area of the vehicle)	1515.4 in ²